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By -

RareBooksClub. Paperback. Book Condition: New. This item is printed on demand. Paperback. 78 pages. Original publisher: Hampton, Va.: National Aeronautics and Space Administration, Langley Research Center; Springfield, Va.: National Technical Information Service, distributor, 1994 OCLC Number: (OCoLC)60677341 Excerpt: . . . As part of thestudy, multipletunnelrunswere Angle of attack. The model angle of attack necessarwyith and without the flow-angle probein is set manually by rotating the turntables to the de-placebecauseof probeinterferencien themeasured sired pitch. This angle is determined from inclinome-ter readings on a reference surface attached to the wall-andslot-pressurdeata. To determinetheremodel turntable. During the experiment, data were peatabilityof the testconditionst, hestandardde-viationof the tunnelMachnumberwasdetermined acquired at 0, 0.5, :t: 1, and 2 on all wall con-figurations. Data were acquired at 4 on some con-from thesameconsecutivreepeatrunsusedfor the figurations. The angles were generally set to within probeanalysis. Thestandarddeviation of the free- 3 rain of arc (J: 0.05). -streamMachnumberwastypicallylessthan0.0017; for freestreamMachnumbersaround0. 7, tile stan Wall-Pressure Data dard deviation of the free-stream Machnumberis about 0.0010. Bothrepeatabilityvaluesareconsid-General observations. Typical wall-pressure eredgoodfor transonic windtunnels. data from the 6 19 Tmmel experiment are shown in figure 28. The tunnelempty (fig. 28 (a)) and Mach number calibrations. Typical tunnel-airfoil-installed (...

Reviews

Very beneficial to all of class of people. I am quite late in start reading this one, but better then never. You may like just how the writer create this publication.

-- Audra Klocko PhD

Thorough information! Its this type of great go through. It is amongst the most incredible publication i actually have read through. It is extremely difficult to leave it before concluding, once you begin to read the book. -- Germaine Welch